

<https://doi.org/10.15407/knit2025.01.017>
UDC 62-932.4

Y. V. IVANCHUK¹, Doctor of Technical Sciences, Prof.

<https://orcid.org/0000-0002-4775-6505>

A. A. YAROVYI¹, Doctor of Technical Sciences, Prof.

<https://orcid.org/0000-0002-6668-2425>

V. V. LIMAN², PhD in Technical Sciences, Associate Prof.

<https://orcid.org/0000-0003-1280-237X>

V. S. OZERANSKYI¹, PhD in Technical Sciences, Associate Prof.

<https://orcid.org/0009-0007-1694-2317>

O. A. KOZLOVSKYI¹, Student

<https://orcid.org/0009-0008-1797-4086>

¹ Vinnytsia National Technical University

95 Khmelnytske Shose Str., Vinnytsia, 21021 Ukraine

² Vinnytsia National Agrarian University

3 Sonyachna Str., Vinnytsia, 21008 Ukraine

MODELING THE STABILITY OF A THROTTLE HYDRAULIC DRIVE OF ROCKET AND SPACE SYSTEMS UNDER RANDOM LOAD AND STOCHASTIC PARAMETERS

The use of a throttle hydraulic drive in rocket and space technology is promising due to its simplicity, reliability in operation, and low metal consumption. It has been determined that vibrations occur in the hydraulic system of rocket and space equipment under external influence. They cause unstable movement of working units, and, as a result, additional vibrations occur on the actuator. Determining the stability of the throttle hydraulic drive is relevant. This will ensure that the rocket and technical system maintains specified equilibrium states or types of motion. The article solves an important scientific and technical problem of increasing the accuracy of identifying the state of a hydraulic drive with throttle control under the action of a stochastic load in rocket and technical systems. A mathematical model of the operation of a hydraulic drive with throttle control is developed based on its calculation scheme. A generalized method for mathematical modeling of the probability of system stability for the mathematical expectation under a random load and in the presence of one random parameter, namely, the modulus of elasticity of the working fluid, is developed. Linearization of viscous friction forces was performed, and the schedule of the standard deviation of the random modulus of elasticity of the working fluid in the Taylor series was used. A solution to the mathematical model in the form of differential equations using a technique based on statistical linearization and expansion in the Taylor series was proposed. In this case, the stability condition of the hydraulic system is determined by the mathematical expectation in the form of the Hurwitz criterion. The stability condition of the hydraulic drive with throttle control is determined based on the probability of system stability, where the value of the random external load is specified in the form of mathematical expectation and variance.

Keywords: hydraulic drive; stochastic parameters; pressure; elastic modulus; stability; rocket and space technology.

Цитування: Ivanchuk Y. V., Yarovyi A. A., Liman V. V., Ozeranskyi V. S., Kozlovskyi O. A. Modeling the stability of a throttle hydraulic drive of rocket and space systems under random load and stochastic parameters. *Space Science and Technology*. 2025. 31, No. 1 (152). P. 17—26. <https://doi.org/10.15407/knit2025.01.017>

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INTRODUCTION

Hydraulic automation equipment is widely applied in the rocket and space industry [12]. In particular, hydraulic lifting mechanisms are exclusively used in the lifting and installation units of space launch complexes to transfer a space rocket from a horizontal to a vertical position [3]. The use of hydraulic systems for controlling various machines and devices in spacecraft testing facilities allows for creating external working load forces with a wide range of dynamic parameter adjustments at the experimental facility [18]. In modern rocket and technical systems with a high degree of cycle automation, it is necessary to implement many different types of motion. A compact hydraulic cylinder is easily built into mechanisms and connected by pipelines to a pump unit. The use of a hydraulic drive opens up wide opportunities for cycle automation, control, and optimization of work processes and is also easily upgraded. Throttle control based on the use of elements with variable flow sections in control circuits is promising for the hydraulic drive of rocket and space technology [9]. These elements regulate the flow rate of the working medium and, if necessary, change the direction of flows. In hydraulic systems with throttle control, the power source provides control circuits for the working environment with a small change in pressure in the pressure line at virtually unlimited costs [16].

The growing requirements for automatic control systems in rocket and space technology have led to an in-depth study of the properties of hydraulic devices and the development of calculation methods for systems built using these devices. It has been determined that vibrations occur in the hydraulic system of rocket and space equipment under external influence. This causes unstable movement of the working units, and, as a consequence, additional vibrations occur on the actuator [10]. In particular, when random vibrations occur, the change in the pressure drop in the cavities of the hydraulic cylinder can be so large that it significantly affects the instantaneous values of the flow rate of the working fluid flowing through the distributor. All this leads to a general decrease in the reliability of rocket and technical systems and, consequently, can result in emergency situations, both on the launch pad and in outer space [10, 22]. Therefore, when designing hydraulic systems, it is necessary to

solve the synthesis problem. It consists of choosing the system structure, its parameters, and the design of the elements in such a way that both resistance to random stochastic loads (vibrations, impacts from space debris) and the necessary quality indicators of the control processes are ensured [7, 18].

The aim of the work is to increase the accuracy of identifying the state of a hydraulic drive with throttle control under the action of a stochastic load in rocket-technical systems by developing and implementing new, more effective methods of mathematical modeling. This will allow for achieving a number of high-quality practical results: increasing the reliability of determining the operating characteristics when designing a hydraulic drive; the possibility of developing rocket-technical systems with improved operational characteristics; and reducing the time for testing rocket and space technologies, etc.

To achieve the set goals, it is necessary to solve the following tasks:

- develop a mathematical model of the operation of a hydraulic drive with throttle control under the action of a stochastic load;
- develop a method for solving a mathematical model of the dynamics of a hydraulic drive with throttle control, which will allow determining the stability of the system under the action of a stochastic load;
- using the developed method, determine the stability of a given motion mode under the action of an external stochastic load.

LITERATURE REVIEW

The work of [22] presents a graph-analytical method for determining the stability region of pressure pulse generators. In this method, unlike existing ones, the Hurwitz stability criterion is used for a linear, non-homogeneous differential equation of the third order to represent the mathematical model of the motion of the shut-off element in the pulsator valve. This allows us to determine the energy relationships of the drive for the occurrence of different types of oscillatory processes.

The practical implementation of such an approach is possible only for mathematical models of mainly low dimensionality and describes the properties of objects under the action of deterministic loads. This restricts the applicability of mathematical mod-

eling results, which do not consider the effects of all transient processes in the hydraulic link [26]. All this results in the accumulation of redundant, unrealized systems of technological movements [6].

In the works of [1, 23], a statistical linearization method was proposed that, unlike the existing ones, uses complex amplitudes and the probability integral. This allows for solving linear non-homogeneous differential equations of the second order of oscillatory systems in stochastic mathematical models of hydraulic systems using the spectral form of amplitude-frequency characteristics. In these models, the functions of the force interaction of the working elements of the pulse drive are a stationary, ordinary, random process. The method used does not take into account the stochastic change in the properties of the working fluid as an element of the energy carrier of the hydraulic system. The consequence of this is the numerical instability (fluctuations) of the solution of the mathematical model in the transient modes of operation of the hydraulic drive.

The physical parameters of the energy carrier (working fluid) and the design characteristics of the hydraulic drive have a significant impact on increasing the speed, energy saturation, and compactness of the hydraulic system. This necessitates the creation of mathematical models as systems of differential equations based on an artificial dynamic model with reduced coefficients [4, 27]. The reduced coefficients reflect the elastic-viscous properties of the hydraulic circuit with subsequent linearization of its dynamic characteristics. This, in turn, leads to the ignoring of wave processes [7] in the hydraulic drive. This approach to modeling does not allow for determining the stability of specified equilibrium states or movements of the hydraulic system.

The state of a hydraulic system can be stable or unstable depending on the characteristics and parameters of the elements that comprise it [21]. Since stability is the ability of a system to maintain specified equilibrium states or ensure specified types of movement, determining stability is, therefore, an urgent task when designing hydraulic systems.

MATERIALS AND METHODS

Figure 1 shows the hydraulic diagram of the hydraulic drive with throttle adjustment [5] for the move-

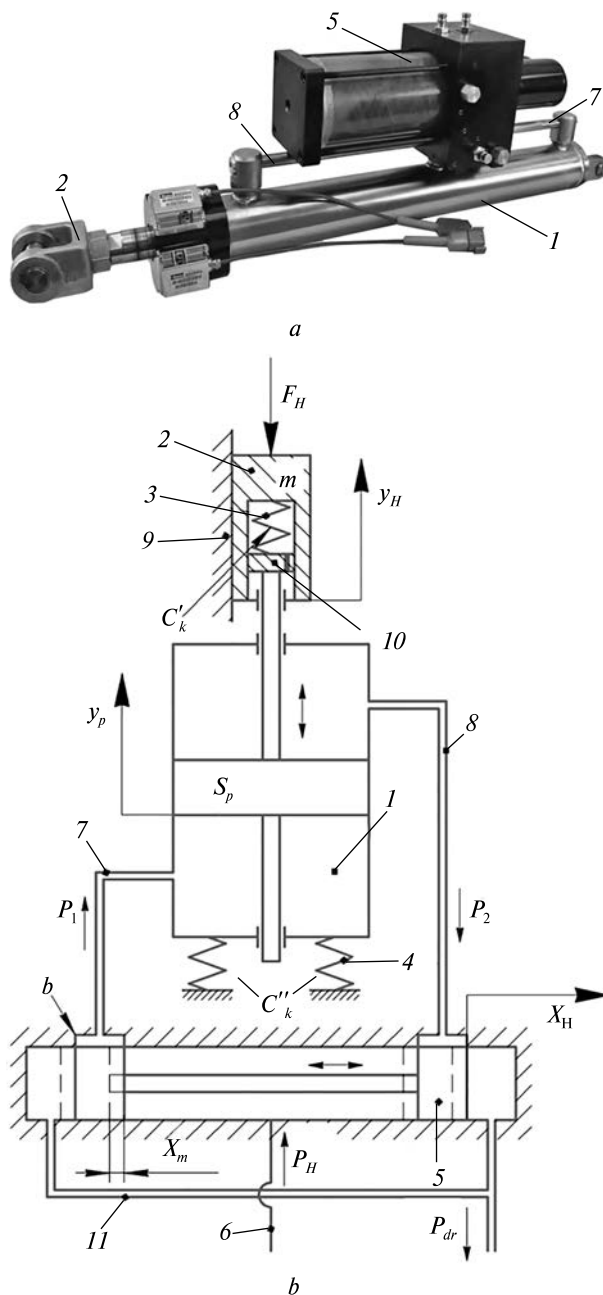


Figure 1. Hydraulic drive with throttle control: *a* — general look, *b* — calculation scheme

ment of the executive body of hydraulic cylinder 1. The hydraulic cylinder rod 1 is connected to the reduced mass m of the external load via a spring 3 with a stiffness of c'_k . The reduced mass 2 contacts the guide 9. A damping piston 10 is placed on the upper

end of hydraulic cylinder rod 1. The hydraulic cylinder 1 is attached to the base via springs 4 with a stiffness of c_k'' . In hydraulic cylinder 1, the working fluid is pumped through hydraulic line 7 by the hydraulic pump via pressure hydraulic line 6 and throttle distributor 5. The working fluid from hydraulic cylinder 1 is drained into the tank via hydraulic line 8 through throttle distributor 5 and drain hydraulic line 11.

The mathematical model of the hydraulic drive with throttle control (see Fig. 1, b) is represented by a system of equations:

$$\begin{aligned} S_p \frac{d}{dt}(y_p - y_c) + \frac{V}{E} \frac{dp_F}{dt} &= Q_1 - r p_F, \\ m \frac{d^2 y_H}{dt^2} &= c_k'(y_p - y_H) + \eta' \frac{d}{dt}(y_p - y_H) - \\ &\quad - \eta \frac{dy_p}{dt} - k_p y_H - F_H - F_{fr} \text{sign} \frac{dy_H}{dt}, \\ p_F S_p &= c_k'(y_p - y_H) + \eta' \frac{d}{dt}(y_p - y_H), \\ -p_F S_p &= c_k''(y_p - y_H) - \eta'' \frac{dy_c}{dt}, \end{aligned} \quad (1)$$

where

$$\begin{aligned} Q_1 &= x_H \mu b \sqrt{\frac{1}{2} |p_H - p_F \text{sign} x| \text{sign} \Delta p}, \\ \Delta p &= p_p - p_F, \quad p_H = p_H - p_{dr}, \quad p_F = p_1 - p_2, \\ x_H &= \begin{cases} x & |x| \leq x_m, \\ x_m \text{sign} x & |x| > x_m, \end{cases} \end{aligned}$$

η' is the coefficient of viscous friction between the rod of hydraulic cylinder 1 and the reduced mass 2, η'' is the coefficient of viscous friction between the piston and the walls of hydraulic cylinder 1, η is the coefficient of viscous damping of the piston of hydraulic cylinder 10 of the reduced mass 2, V is the volume of the pressure cavity of the hydraulic drive, S_p is the area of the piston of hydraulic cylinder 1, r is the loss coefficient of the hydraulic drive, k_p is the coefficient of positional load on the initial link 2 of the hydraulic drive, p_p is the difference between the supply pressure p_H from the hydraulic pump and the drain pressure p_{dr} , p_F is the difference in pressure on the piston of hydraulic cylinder 1, y_H is the displacement of the reduced mass 2, y_p is the displacement of the piston of hydraulic cylinder 1, E is the reduced modulus of elasticity (random variable), F_H

is the external load on reduced mass 2 (random stationary function), F_{fr} is the dry friction force between the reduced mass 2 and guide 9, b is the width of the throttle opening of hydraulic distributor 5, x_H is the displacement of the piston of throttle hydraulic distributor 5, x_m is the maximum displacement of the piston of throttle hydraulic distributor 5, p_1 is the discharge pressure of hydraulic pump 1, p_2 is the drain pressure of hydraulic pump 1, μ is the flow coefficient of throttle hydraulic distributor 5.

Taking the external disturbance as $x = 0$, the system of equations (1) can be written in the following form [17]:

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\eta_{fr}(\sigma) + \mathbf{B}F_H, \quad s = x_2, \quad (2)$$

where

$$\mathbf{A} = \begin{bmatrix} 0 & 1 & 0 & 0 & 0 \\ -a_2 & -a_3 & a_4 & 0 & a_1 \\ -\gamma & -\theta_1 & -\beta & \theta & \gamma \\ 0 & 0 & -a & -b & 0 \\ c & 1 & e & 0 & -c \end{bmatrix},$$

$$\mathbf{B} = \begin{bmatrix} 0 \\ 1 \\ m \\ 0 \\ 0 \\ 0 \end{bmatrix}, \quad \mathbf{x} = \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \\ x_5 \end{bmatrix},$$

$$a = \frac{S_p}{\eta''}, \quad b = \frac{c_k''}{\eta''}, \quad c = \frac{c_k'}{\eta'}, \quad e = \frac{S_H}{\eta'},$$

$$a_1 = \frac{c_k' - c\eta'}{m}, \quad a_2 = \frac{c_k' - c\eta' + k_p}{m},$$

$$a_3 = \frac{F_{fr}}{m}, \quad a_4 = \frac{e\eta'}{m},$$

$$\alpha = \frac{2E}{V}, \quad \beta = \alpha r + \alpha(e + a)S_p,$$

$$\theta_1 = \alpha S_p, \quad \gamma = \theta_1 c, \quad \theta = \theta_1 c,$$

$$x_1 = y_H, \quad x_2 = \frac{dy_H}{dt}, \quad x_3 = p_F,$$

$$x_4 = y_c, \quad x_5 = y_p.$$

It is necessary to determine the probability of system stability by the mathematical expectation under random load and in the presence of one random pa-

parameter. Let us assume that the load $F_H(t)$ is a random stationary function with a known mathematical expectation m_{PH} , variance D_{PH} , and correlation function $K_{PH}(\tau)$.

The reduced modulus of elasticity of the system E is considered a random parameter. To justify this assumption, it is necessary to note that the modulus of elasticity E depends on a number of factors: pressure and rate of pressure change [15], gas-air phase content [22], temperature of the air working fluid, etc. Considering that these factors are independent of each other and some of them are random variables, the value of the modulus of elasticity of the working fluid E is taken as a random variable with a normal probability density distribution and a given mathematical expectation m_E and variance D_E .

To solve the problem, it is necessary to apply an approach based on the statistical linearization method [2]. When $m_{FH} = \text{const}$, the mathematical expectation $m_x(t)$ in the established mode is determined from a system of nonlinear algebraic equations [20]:

$$\begin{aligned} m_{x_2} &= 0, \\ -a_2 m_{x_1} - (a_3 + k_0/m) m_{x_2} + a_4 m_{x_3} + \\ &+ a_1 m_{x_5} = \frac{m_{FH}}{m}, \\ -\gamma m_{x_1} - \theta m_{x_2} - \beta m_{x_3} - \theta m_{x_4} + \gamma m_{x_5} &= 0, \\ -a m_{x_3} - b m_{x_4} &= 0, \\ c m_{x_1} + m_{x_2} + e m_{x_3} - c m_{x_5} &= 0, \end{aligned} \quad (3)$$

where

$$\begin{aligned} k_0(0, \sigma_{x_2}) &= k_1(0, \sigma_{x_2}) = \\ &= \eta_2 + \frac{2l}{\sigma_{x_2} \sqrt{2\pi}} - 2(\eta_1 + \eta_2) \Phi\left(\frac{\varepsilon}{\sigma_{x_2}}\right), \end{aligned}$$

and parameters l_1 , $\eta_1 = \text{tg} \eta'$, $\eta_2 = \text{tg} \eta''$ are known values determined from the nonlinear characteristic diagram Figure 2.

The solution of system (3) is

$$m_x = \alpha = \left| -\frac{m_{PH}}{m}, 0, 0, 0, -\frac{m_{PH}}{m} \right|.$$

Taking into account:

$$\beta = \text{grad} k_0(m_x, \sigma_x) \Big|_{m_x} = \left| 0, \frac{\partial k_0}{\partial m_{x_2}} \Big|_{m_x}, 0, 0, 0 \right|$$

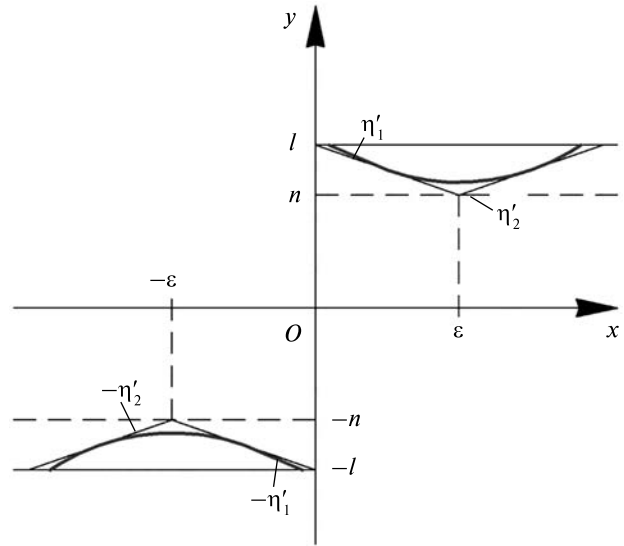


Figure 2. Diagram of the characteristics of a nonlinear element

and B , c^* , α , $\beta^* = 0$, we have the Hurwitz stability condition [24] for the following polynomial [4, 7]:

$$\begin{aligned} \Delta &= |pE - A - Bk_0(0, \sigma_{x_2})c^*| = \\ &= p^5 + d_1 p^4 + d_2 p^3 + d_3 p^2 + d_4 p + d_5, \end{aligned} \quad (4)$$

where

$$\begin{aligned} d_1 &= \mu_2 + a_3, \\ d_2 &= \mu_1 + a_3 \mu_2 + a_2 + a_4 \theta_1 - a_1, \\ d_3 &= \mu_0 + a_3 \mu_1 + (a_2 - a_1) \mu_2 + a_4 \theta_1 (b + c) + a_1 \theta c, \\ d_4 &= a_3 \mu_0 + (a_2 - a_1) \mu_1 + \theta b (a_4 c + a_1 e), \\ d_5 &= (a_2 - a_1) \mu_0, \\ \mu_0 &= b \beta c - a \theta c - \gamma e b, \\ \mu_2 &= b + \beta + c, \\ \mu_1 &= b \beta + c b + c \beta - a \theta - \gamma e, \\ a_3 &\equiv a_3 + k_0/m. \end{aligned}$$

The stability condition according to Hurwitz [7, 24] for Δ :

$$d_1 > 0, \delta_2 > 0, \delta_3 > 0, \delta_4 > 0, d_5 > 0, \quad (5)$$

where

$$\begin{aligned} \delta_2 &= d_1 d_2 - d_3, \quad \delta_3 = d_3 \delta_2 - d_1 \delta_0, \\ \delta_4 &= d_4 \delta_0 - d_2 d_5 \delta_2 - d_5 \delta_0, \quad \delta_0 = d_1 d_4 - d_5. \end{aligned}$$

In addition to condition (5), the equation is used:

$$\sigma_{x_2}^2 = \int_{-\infty}^{+\infty} \left| \frac{L(i\omega)}{\Delta(i\omega)} \right| S_{p_H}(\omega) d\omega, \quad (6)$$

where $S_{p_H}(\omega)$ — special density of a stationary random function,

$$L(p) = -p(p^3 + \mu_2 p^2 + \mu_1 p + \mu_0) / m.$$

To solve equation (6) after integration, it is necessary to write it in implicit form:

$$\Phi_1(k_1(0, \sigma_{x_2}), \sigma_{x_2}(E)) = 0, \quad (7)$$

where from

$$\Phi_1(k_1(0, \sigma_{x_2}), \sigma_{x_2}(m_E)) = 0. \quad (8)$$

Then, in the region of the point, m_E is determined by Taylor's formula [11]:

$$\begin{aligned} \sigma_{x_2}(E) &= \sigma_{x_2}(m_E) + \frac{\partial \sigma_{x_2}(m_E)}{\partial E} (E - m_E) + \\ &+ \frac{1}{2} \frac{\partial^2 \sigma_{x_2}(m_E)}{\partial E^2} (E - m_E)^2 + R_3, \end{aligned} \quad (9)$$

where $\sigma_{x_2}(m_E)$ is determined from equation (8) and

$$\begin{aligned} \frac{\partial \sigma_{x_2}(m_E)}{\partial E} &= - \frac{\partial \Phi_1 / \partial E}{\partial \Phi_1 / \partial \sigma_{x_2}} \Big|_{E=m_E} \\ \frac{\partial^2 \sigma_{x_2}(m_E)}{\partial E^2} &= \frac{\partial}{\partial E} \left(\frac{\partial \Phi_1 / \partial E}{\partial \Phi_1 / \partial \sigma_{x_2}} \right) \Big|_{E=m_E}, \end{aligned}$$

R_3 — remainder term of the equation.

After determination $\sigma_{x_2} = \beta(E)$

$$\Sigma_{x_2} = M[\sigma_{x_2}] = \int_{-\infty}^{+\infty} \beta(E) f(E) dE, \quad (10)$$

$$\delta^2 = D[\sigma_{x_2}] = \int_{-\infty}^{+\infty} \beta^2(E) f(E) dE - \Sigma_{x_2}^2$$

and the stability region D based on inequalities (5), the probability of system stability is determined as:

$$P = \int_D f(E) dE,$$

where

$$f(E) = \frac{1}{\sqrt{2\sigma_E}} \exp\left(-\frac{(E - m_E)^2}{2\sigma_E^2}\right).$$

RESULTS AND DISCUSSION

To implement the developed method of mathematical modeling of the stability of a throttle hydraulic drive (see Fig. 1, b), the following is adopted: $\eta' = \eta'' = 0$ N·s²/m, $c'_k = \infty$ N/m, $k_p = 0$ N/m, $c''_k = 51.8 \cdot 10^5$ N/m, $m = 5 \cdot 10^3$ kg, $\eta = 2.34 \times$

$\times 10^4$ N·s²/m, $V = 7.05 \cdot 10^{-3}$ m³, $S_p = 78.54 \cdot 10^{-4}$ m², $r = 2 \cdot 10^{-11}$ m⁵/(N·s).

Also, for the random modulus of elasticity, the value [2] is specified in the form of the mathematical expectation $m_E = 0.8 \cdot 10^9$ N/m², the standard deviation $\sigma_E = 0.07 \cdot 10^9$ N/m², and the probability density function is determined by the following formula:

$$f(E) = \frac{1}{\sqrt{2\pi}\sigma_E} \exp\left(-\frac{(E - m_E)^2}{2\sigma_E^2}\right).$$

For a random external load, the value [11] is also specified as a mathematical expectation $m_{p_H} = 3.0 \cdot 10^9$ N, a variance $D_{p_H} = 45.0 \cdot 10^6$ N², and the following formula determines the correlation function:

$$K_{p_H}(\tau) = D_{p_H} \exp\left(-\frac{|\tau|}{0.62}\right).$$

The diagram of the characteristics of the external nonlinear load of viscous friction forces acting on hydraulic cylinder piston 1 (see Fig. 1) is shown in Figure 3.

For the given input data based on equations (2):

$$\begin{aligned} \mathbf{A} &= \begin{bmatrix} -b & c \\ -\alpha & -\beta \end{bmatrix}, \\ \mathbf{x} &= \begin{bmatrix} x_2 \\ x_3 \end{bmatrix}, \quad \mathbf{B} = \begin{bmatrix} -1/m \\ 0 \end{bmatrix}, \quad \mathbf{c} = \begin{bmatrix} 1 \\ 0 \end{bmatrix}, \\ b &= \frac{\eta}{m} = \frac{2.34 \cdot 10^4}{5 \cdot 10^3} = 4.68 \text{ N·s}^2/(\text{m·kg}), \\ \alpha &= \frac{S_p}{d}, \quad c = \frac{S_p}{m} = \frac{78.54 \cdot 10^{-4}}{5 \cdot 10^3} = 15.708 \cdot 10^{-7} \text{ m}^2/\text{kg}, \\ \beta &= \frac{r}{d}. \end{aligned}$$

Taking $m_{x_2} = 0$, we obtain the conditional resistance according to Hurwitz in the form of [7, 20]:

$$\begin{aligned} d_1 &= k_2 + b + \beta > 0, \\ d_2 &= k_2\beta + b\beta + \alpha c, \end{aligned} \quad (11)$$

where $k_2 \equiv k_0(0, \sigma_{x_2}) / m = k_1(0, \sigma_{x_2}) / m$.

Calculating the spectral density $S_{p_H}(\omega)$ through the correlation function $K_{p_H}(\tau)$ from equation (6), we obtain the following:

$$\begin{aligned} \sigma_{x_2}^2 &= \frac{\sigma_{p_H}^2}{0.62\pi m} \int_{-\infty}^{+\infty} \left| \frac{i\omega + \beta}{(i\omega)^3 + i\omega d_1 + d_2} \right|^2 \frac{d\omega}{(1.6 + i\omega)^2} = \\ &= \frac{2\sigma_{p_H}^2}{0.62m^2} J^3, \end{aligned}$$

where

$$J = \frac{1}{2\pi} \int_{-\infty}^{+\infty} \left| \frac{g_3(i\omega)}{h_3(i\omega)h_3(-i\omega)} \right|^2 d\omega,$$

$$g_3(z) = \beta^2 - z^2,$$

$$h(z) = z^3 + (d_1 + 1.6)z^2 + (d + 1.6d_1)z + 1.6d_2.$$

Finally, we get:

$$\begin{aligned} \Phi_1(k_2, \sigma_{x_2}, E) &\equiv \sigma_{x_2}^2 (k_2 + b + \beta)(k_2\beta + b\beta + \alpha c) \times \\ &\times [0.62(k_2\beta + b\beta + \alpha c) + k_2 + b + \beta + 1.6] - \\ &- 1.8[k_2\beta + b\beta + \alpha c + 0.62\beta^2(k_2 + b + \beta + 1.6)] = 0, \end{aligned}$$

where

$$k_2 = \frac{k_1(0, \sigma_{x_2})}{m} = \frac{0.191}{\sigma_{x_2}} - 0.4\Phi\left(\frac{0.2}{\sigma_{x_2}}\right).$$

From the equation $\Phi_1(k_2, \sigma_{x_2}, m_E) = 0$, we determine $\sigma_{x_2}(m_E) = 1.37 \cdot 10^{-2} \text{ N/m}^2$. Based on expression (9), we determine:

$$\begin{aligned} \sigma_{x_2}(E) &= 1.37 \cdot 10^{-2} - 1.8 \cdot 10^{-11} (E - m_E) - \\ &- \frac{1}{2} 0.355 \cdot 10^{-20} (E - m_E)^2. \end{aligned} \quad (12)$$

Using equations (10) and (12), we can obtain that:

$$\Sigma_{x_2} = 1.369 \cdot 10^{-2}, \quad \delta^2 = 1.894 \cdot 10^{-2}.$$

Then, based on Chebyshev's inequality [8, 26], we can obtain:

$$\begin{aligned} 0.8439 \cdot 10^{-2} &= \Sigma_{x_2} - 3\delta < \sigma_{x_2} < \Sigma_{x_2} + 3\delta = \\ &= 1.894 \cdot 10^{-2}. \end{aligned} \quad (13)$$

The stability region D of the system is determined based on equation (11). To do this, it is necessary to approximate the function $k_2(0, \sigma_{x_2})$ by a linear function in the region of the point $M[\sigma_{x_2}] = 1.369 \cdot 10^{-2}$:

$$k(0, \sigma_{x_2}) \approx -1019 + 27.7\sigma_{x_2}. \quad (14)$$

Substituting equation (14) into (11) and taking into account (12), we obtain:

$$D\{0 \leq E < -2.9 \cdot 10^8\}.$$

Then the probability of stability of the system is:

$$\begin{aligned} P &= \int_D f(E) dE = \\ &= 1 - \frac{1}{\sqrt{2\pi D_E}} \int_{-2.9 \cdot 10^8}^0 \exp\left[-\frac{(E - m_E)^2}{2D_E}\right] dE = \\ &= 1 - \Phi(0.47) = 0.82. \end{aligned}$$

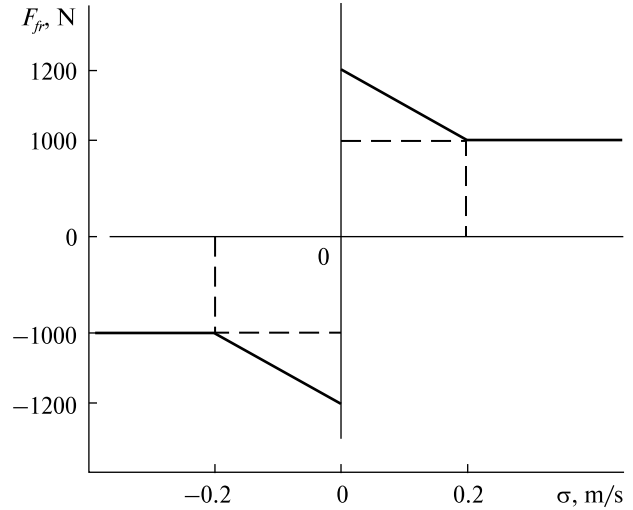


Figure 3. Diagram of the characteristics of the nonlinear element of the piston's viscous friction forces

Thus, a solution was obtained that provides a quantitative assessment of the probability of stability of the hydraulic system in Figure 1. The possibility of meeting the stability conditions of the hydraulic drive depends on the requirements for its speed. It also depends on the degree of damping created by friction forces both in the hydraulic cylinder and the load and on the permissible flow rate of fluid from the power supply system at the equilibrium state of the hydraulic drive [14]. If it is necessary to obtain a high-quality factor of the hydraulic drive in the presence of a large reduced mass m of the external load to the hydraulic cylinder rod and at small values of the viscous damping coefficient η , then additional measures are used to ensure stability. They are as follows: an overflow of working fluid is introduced between the cavities of hydraulic cylinder 1, an elastic element 4 of hydraulic cylinder support 1 with a rigidity of c_k'' is used, and a damper is installed on the spool of the throttle distributor [19, 25].

CONCLUSIONS

Based on the theoretical studies performed, an important scientific and technical problem of increasing the accuracy of identifying the state of a hydraulic drive with throttle control under the action of a stochastic load in rocket and spacecraft systems was solved. This was achieved by developing a generalized

method for mathematical modeling of the probability of system stability according to the mathematical expectation under random load and in the presence of one stochastic parameter.

A solution to the mathematical model in the form of differential equations is proposed using a technique based on statistical linearization through expansion in a Taylor series. In this case, the stability condition of the hydraulic system is determined

by the mathematical expectation in the form of the Hurwitz criterion.

Using the developed method, the stability condition of a hydraulic drive with throttle control is determined. In this case, for a random external load, the value of which is specified as a mathematical expectation $m_{p_H} = 3.0 \cdot 10^9$ N and dispersion $D_{p_H} = 45.0 \cdot 10^6$ N², the probability of system stability is equal to $P = 0.82$.

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Стаття надійшла до редакції 26.10.2024

Після доопрацювання 10.11.2024

Прийнято до друку 20.11.2024

Received 26.10.2024

Revised 10.11.2024

Accepted 20.11.2024

Я. В. Іванчук¹, д-р техн. наук, проф.
<https://orcid.org/0000-0002-4775-6505>

А. А. Яровий¹, д-р техн. наук, проф.
<https://orcid.org/0000-0002-6668-2425>

В. В. Ліман², канд. техн. наук, доцент
<https://orcid.org/0000-0003-1280-237X>

В. С. Озеранський¹, канд. техн. наук, доцент
<https://orcid.org/0009-0007-1694-2317>

О. А. Козловський¹, студент
<https://orcid.org/0009-0008-1797-4086>

¹ Вінницький національний технічний університет
Хмельницьке шосе 95, м. Вінниця, Україна, 21021

² Вінницький національний аграрний університет
вул. Сонячна 3, м. Вінниця, Україна, 21008

МОДЕЛЮВАННЯ СТІЙКОСТІ ДРОСЕЛЬНОГО ГІДРОПРИВОДУ РАКЕТНО-КОСМІЧНИХ СИСТЕМ ПРИ ВИПАДКОВОМУ НАВАНТАЖЕННІ ТА СТОХАСТИЧНИХ ПАРАМЕТРАХ

Перспективним є застосування в ракетно-космічній техніці дросельного гідроприводу завдяки своїй простоті, надійності в експлуатації та невисокій металомісткості. При зовнішньому впливі в гідросистемі ракетно-космічного обладнання виникають вібрації, які призводять до нестабільного руху робочих вузлів, внаслідок чого виникають додаткові коливання на виконавчому органі. Актуальним є визначення умов стійкості роботи дросельного гідроприводу, що дозволить ракетно-технічній системі зберігати задані рівноважні стани або види руху. У роботі вирішено важливу науково-технічну проблему підвищення точності ідентифікації стану гідроприводу із дросельним регулюванням при дії стохастичного навантаження у ракетно-технічних системах. Розроблено математичну модель роботи гідроприводу із дросельним регулюванням на основі її розрахункової схеми. Розроблено узагальнений

метод математичного моделювання ймовірності стійкості системи по математичному сподіванню при випадковому навантаженні і при наявності одного випадкового параметра, а саме модуля пружності робочої рідини. Було проведено лінеаризацію сил в'язкого тертя і використано розклад в ряд Тейлора значень стандартного відхилення параметра випадкового модуля пружності робочої рідини. Запропоновано розв'язок математичної моделі у вигляді диференціальних рівнянь із використанням методики на основі статистичної лінеаризації на основі розкладу в ряд Тейлора, де умова стійкості гідросистеми визначається за математичним сподіванням у вигляді критерію Гурвіца. Визначено умову стійкості гідроприводу із дросельним регулюванням на основі ймовірності стійкості системи, де значення випадкового зовнішнього навантаження задано у вигляді математичного сподівання і дисперсії.

Ключові слова: гідропривод, стохастичні параметри, тиск, модуль пружності, стійкість, ракетно-космічна техніка.